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NAVAL AIR ENGINEERING CENTER PHILADELPHIA, PA. 19112

AERONAUTICAL MATERIALS IABORATORY

REPORT NO. NAEC-AML-2083

DATE 30 Nov 1964

INVESTIGATION INTO THE ELECTRICAL CONDUCTIVITY AND MECHANICAL PROPERTIES OF ALUMINUM ALLOYS SUBJECTED TO ELEVATED TEMPERATURE EXPOSURE

FROBLEM ASSIGNMENT NO. 10-40 UNDER BUREAU OF NAVAL WEAPONS WEPTASK RRMA 02 018/200 1/R007 05 01

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TABLE OF CONTENTS

	PAGE
ABSTRACT	3
INTRODUCTION	Ĩ ₄
EXPERIMENTAL PROCEDURE	4
results and analysis	5
CONCLUSIONS	5
RECOMMENDATIONS	6

LIST OF TABLES

- 1 Results of Test
- 2 Conductivity Data for Various Aluminum Alloys

LIST OF PLATES

1 - Electrical Conductivity Measuring Technique
2 - Conductivity vs Yield Strength for Aluminum Alloy 6061-T6
3 - Conductivity vs Yield Strength for Aluminum Alloy 2024-T81
4 - Conductivity vs Yield Strength for Aluminum Alloy 2219-T81
5 - Conductivity vs Yield Strength for Aluminum Alloy 7075-T73
6 - Conductivity vs Yield Strength for Aluminum Alloy 2020-T6
7 - Conductivity vs Yield Strength for Aluminum Alloy 7075-T6
8 - Conductivity vs Yield Strength for Aluminum Alloy 7178-T6
9 - Conductivity vs Yield Strength for Aluminum Alloy 7002-T6
10 - Conductivity vs Yield Strength for Aluminum Alloy 7079-T6
11 - Conductivity vs Yield Strength for Aluminum Alloy 2024-T3

ABSTRACT

The relationship between the electrical conductivity (as measured by Magnatester Conductivity Meter, Floo Series) and strength properties of bare aluminum alloys 7075-T6, 7075-T73, 6061-T6, 7178-T6, 7002-T6, 2024-T81, 2024-T3, 7079-T6, 2020-T6, and 2219-T81 was investigated in an attempt to correlate conductivity with heat damage to aircraft structural alloys.

I. INTRODUCTION

The Bureau of Naval Weapons has expressed concern over heat damage to aircraft structures in locations subject to power plant heating. Since this heat damage occurs at temperatures below that which causes changes in the paint surface appearance, it is necessary to use some other parameter to assess the damage. A method which measures the electrical conductivity of the alloy by means of eddy current techniques has been suggested and is in fact being used to assess damage to aluminum alloy 7075-T6 in the A-4 aircraft. It is the purpose of this investigation to obtain a correlation between conductivity as measured by the Magnatester FlOO Series and strength properties of other aluminum alloys.

II. EXPERIMENTAL PROCEDURE

The bare aluminum alloys used in this investigation were 7075-T6, 7075-T73, 6061-T6, 7178-T6, 7002-T6, 2024-T81, 2024-T3, 7079-T6, 2020-T6, 2219-T81.

Tensile test specimens of the above alloys were manufactured from available material. Most of the alloys were in the form of bare sheet of various thicknesses. Several alloys, however, were available only as plate from which round tensile specimens were machined. The form of the test specimens used for each alloy is given with the results in Table 1. For each alloy, the test specimens all had the same orientation with relation to the rolling direction.

Conductivity and hardness measurements were made on each material in the as-received condition. The test specimens were then exposed for various times at temperatures of 400°F, 500°F, and 600°F. Conductivity and hardness measurements were then made on the exposed specimens. Following this, the specimens were tensile tested at a strain rate of approximately 0.035 in/in/min. and the 0.2% yield strength and ultimate tensile strength were calculated.

The conductivity measurements were made using a Magnatester Conductivity Meter (FlOO Series), Plate 1. The sensitivity of this instrument was $\pm 0.3\%$ IACS and the accuracy is within $\pm 1-1/2\%$ of scale reading. During calibration of the Magnatester, the standard calibration specimens furnished with the instrument were maintained at the same temperature as the specimens whose conductivity was to be measured, thus eliminating a temperature induced error.

III. RESULTS AND ANALYSIS

The results obtained in this investigation are given in Table 1. Since the yield strength is the most useful of the properties investigated, it is plotted as a function of the conductivity for the various alloys in Plates 1 to 10. Table 2 gives the value of the conductivity and amount of change in the conductivity which corresponds to a 20% loss in yield strength. The amount of change is a measure of the ability of the Conductivity Meter to detect a loss in strength.

From the data in Table 2 and Plates 2, 3, 4, 5, and 6, it is apparent that no useful correlation between conductivity and strength properties is possible for alloys 6061-T6, 2024-T81, 2219-81, 7075-T73, and 2020-T6. In each of these cases, the change in conductivity between the original material and that which has been severely reduced in strength properties by heating is so small that it must be considered non-reproducible.

Alloys 7075-T6, 7178-T6, 7002-T6, and 7079-T6 give useful correlations between conductivity and yield strength as can be seen from Table 2 and Plates 7, 8, 9, and 10.

While alloy 2024-T3, Plate 11, has a very large change in conductivity for a 20% loss in yield strength, it is improbable that a useful correlation can be obtained since only at a high conductivity is a significant loss in strength observed and this loss occurs over a very small conductivity range. A lower conductivity limit could be chosen to eliminate all damaged structures, but in view of the shape of the curve, it could cause a replacement of a large number of structures whose strength was not significantly reduced.

IV. CONCLUSIONS

- 1. Heat damage to structures can be correlated to conductivity readings obtained using a Magnatester Conductivity Meter (F100 Series) for aluminum alloys 7075-T6, 7178-T6, 7002-T6, and 7079-T6.
- 2. A similar correlation cannot be made for aluminum alloys 6061-T6, 2024-T81, 7075-T73, 2219-T81, and 2020-T6 due to the small change in conductivity which occurs due to heat damage.
- 3. The conductivity of alloy 2024-T3 can be correlated, but such a correlation would lead to rejects of many heated but unweakened structures along with those which had been damaged. Above a conductivity of 38%, a high probability exists that this material has been weakened to below the as-received strength and tests of other types should be made.

V. RECOMMENDATIONS

It is recommended that the Magnatester Conductivity Meter (Series F100) be considered for use in detecting heat damage to aluminum alloys 7178-T6, 7002-T6, and 7079-T6 in addition to 7075-T6 alloy

			RESULTS OF TEST			
Alloy & Form	Exposure Temperature	Exposure Time	Conductivity Range, % IACS	Hardness	0.2% Yield Strength	Ultimate Strength
7075-T73	As-Received	ived	37.5	87 02	63 500	000 37
0.065" Sheet	400°F	l hr.	38.5	84 /S	200, 88	07,27
	400°F	5 hrs.	40.5		43,800	59.500
	510°F	10 min.	41.5	61 RR	34,600	53,100
	4006F	l hr.	41.0	65 RB	37,400	55,800
	600°F	_	39.5	$31 R_{ m B}$	19,500	43,900
	3009	l hr.	41.5	75 RF	18,900	41,700
6061-T6	As-Rece	eceived	39.0	56 RB	40.900	46.700
0.063" Sheet	4000F	1 hr.	40.0	54 RB	40,000	46,700
	400°F	5 hrs.	40.5		40,800	44,000
	510°F	10 min.	41.0	33 RB	33,000	38,300
	4006F	l hr.	41.0	85 RF	35,000	41,200
	600°F	10 min.	42.3	;	18,700	28,600
	600°F	l hr.	42.5	58 RF	16,700	27,700
7178-T6	As-Rece	eceived	29.5	95 RB	84,200	91,800
0.100" Sheet	400of	l hr.	36.0	92 RB	78,800	85,200
	4000F	5 hrs.	41.0	75 RB	48,000	62,800
	510°F	10 min.	40.5	66 RB	37,400	55,900
	40067	l hr.	41.3	70 RB	41,800	58,700
	40009	10 min.	40.5	40 RB	21,600	45,500
	600°F	l hr.	41.8	80 RF	20,000	42,200
2020-T6	As-Received	ived	20.5	91 RB	73,500	78 666
0.064" Sheet	4000F	1 hr.	20.5		71,600	76,757
	4000£	5 hrs.	21.0		67,968	73,906
	500°F	l hr.	21.9	74 RB	46,400	58,787
	₹0009	l hr.	23.3		28,100	7,000

TABLE 1 PAGE 1 OF 3 PAGES

Alloy & Form	Exposure Temperature	Exposure	Conductivity Range, 7 IACS	Hardness	0.2% Yield Strength	Ultimate <u>Strength</u>
7079. T6		Received	\prec		70,000	80,000
0.088" Sheet	3c007	10 min.	3		62,500	74,300
	4000F	l hr.	34.0		62,500	74,200
	4000F	5 hrs.	9		44,600	61,500
	510°F	10 min.	35.8	63 RB	34,700	26,600
	490 _° F	l hr.	36.5		37,900	58,200
	600°F	10 min.	9		21,200	48,000
	40009	1 hr.	37.0		20,800	45,000
7075-T6	As-Rece	Received	32.0	93 R _B	79,000	85,500
0.100" Sheet	4000F	l hr.	37.0		72,400	79,600
	4000F	5 hrs.	41.0	70 RB	45,000	59,700
	510°F	10 min.	41.5		35,500	54,300
	490cF	l hr.			39,200	26,600
	€000F	10 min.	41.5		20,000	44,200
	3009	l hr.	42.0	76 RF	18,600	41,100
2219-T81		Received	2	77 RB	52,100	67,290
0.062" Sheet			32.0		52,100	68,700
	400°F		2.		34,900	46,500
	4000F	5 hrs.	2.		35,500	46,600
	500°F	l hr.	33.5	68 R _R	40,800	59,200
	600°F	l hr.	4.	58 RB	33,200	51,000
7002-T6	As-Rece	Received	33.5	82 RB	58,900	69,700
0.062" Sheet	4000F	lo min.	•	76 R _B	54,200	65,500
	4000F	l hr.	36.0		54,100	66,800
	510°F	10 min.	•	49 R _B	29,600	50,200
	490oF	l hr.	•		32,600	52,200
	600°F	10 min.	•	26 RB	17,300	41,800
	40009	l hr.	38.5	73 RF	16,600	41,200

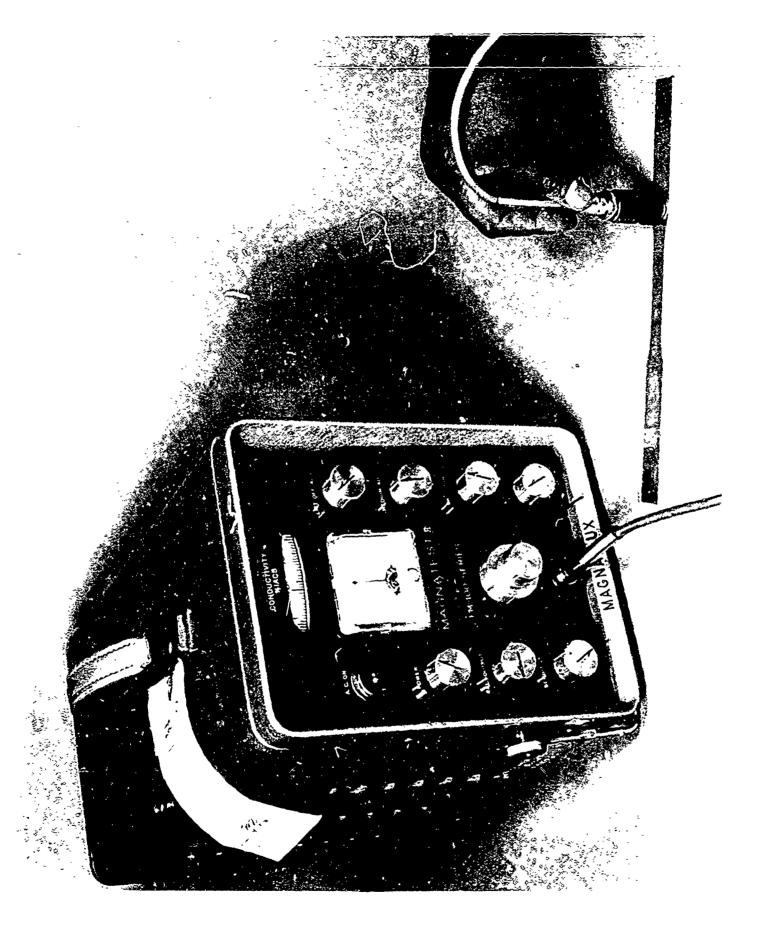
TABLE 1
PAGE 2 OF 3 PAGES

Ultimate Strength	72,100	72,000	70,800	67,100	67,100	57,800	54,400		009,69	67,200	65,000	68,400	56,500	57,400	76 700	42,400	39,600
0.2% Yield Strength	99,400	66,300	63,800	26,600	57,500	40,500	36,600	007 97	47,400	43,000	56,100	60,300	41,000	40,800	26, 600	20,000	15,900
Hardness							$61 R_{B}^{2}$						64 RH		ָ ֭֭֭֭֭֓֞֜֞		1
Conductivity Range, % IACS	38.0	37.5	38.7	38.8	39.5	0.04	41.0	0.00	0 (30.0	38.5	38.8	40.3	41.5	42.5	1 0	42.5
Exposure Time	Received	l hr.	5 hrs.	IO min.	I hr.	lo min.	l hr.	lecefved		l nr.	LO min.	l hr.	lo min.	l hr.	3 hrs.		z nrs.
Exposure Temperature		4000 k	4000F	2000	4900F	600°F	40009	As-Rece	4000	# 00 t	510°F	4006t	4000 4000	# 000 # 000	4009	7000F	
Alloy & Form	2024-T81	o.127 Sheet						2024-T3	0 125" Shoot	מיזיה חוופפר							

TABLE 1 PAGE 3 OF 3 PAGES

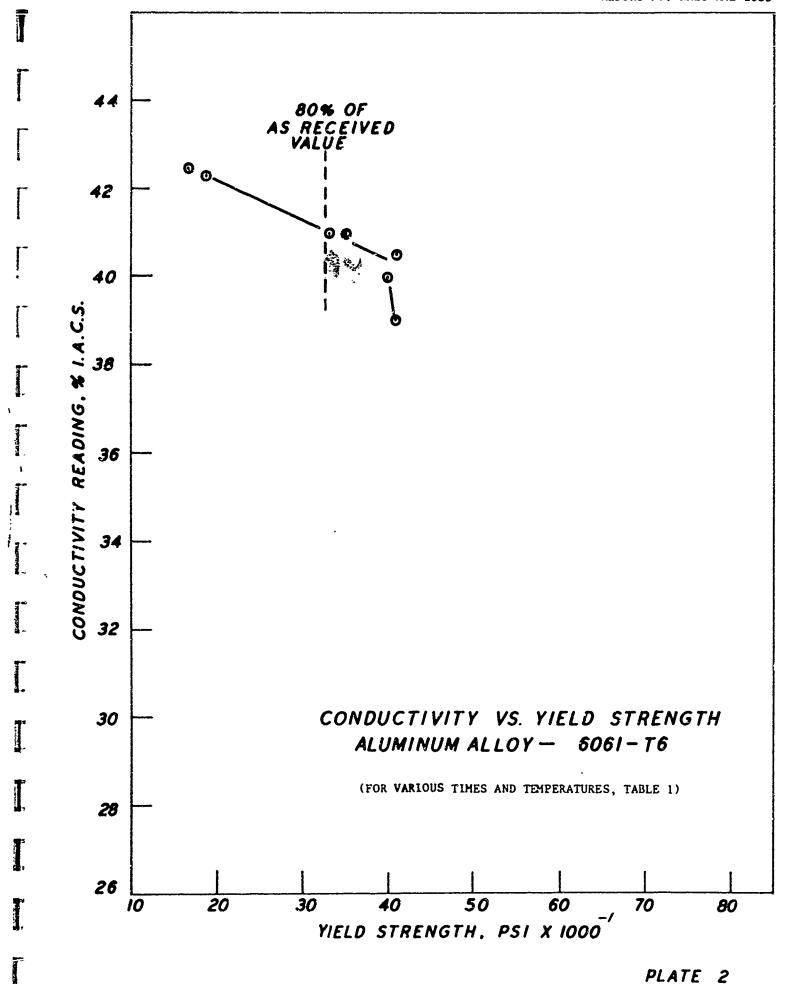
CONDUCTIVITY DATA FOR VARIOUS ALUMINUM ALLOYS

	Alloy	Change in Conductivity, % IACS for 20% Loss of Yield Strength	Conductivity at 80% As-Received Yield Strength
r e	2219 - T81	ı	33.5
	7002 - T6	3.1	36.6
	7178-16	8.0	37•5
<u>~</u>	2024-181	2.0	39•5
	7075- T 73	2.0	39•5
•	6061-Т6	2.0	41.0
	7079 - T6	3•5	34.5
	2024 - T3	12.0	41.0
! ~	7075 - T6	6.0	38.0
	20 20- T6	1.0	21.5



ELECTRICAL CONDUCTIVITY MEASURING TECHNIQUE

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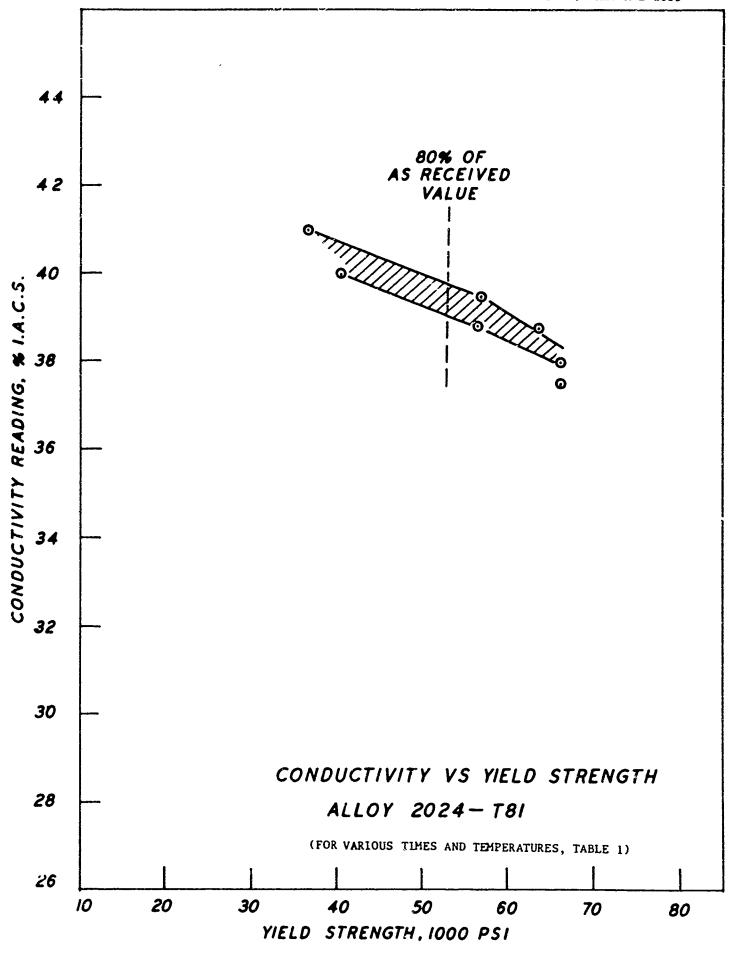


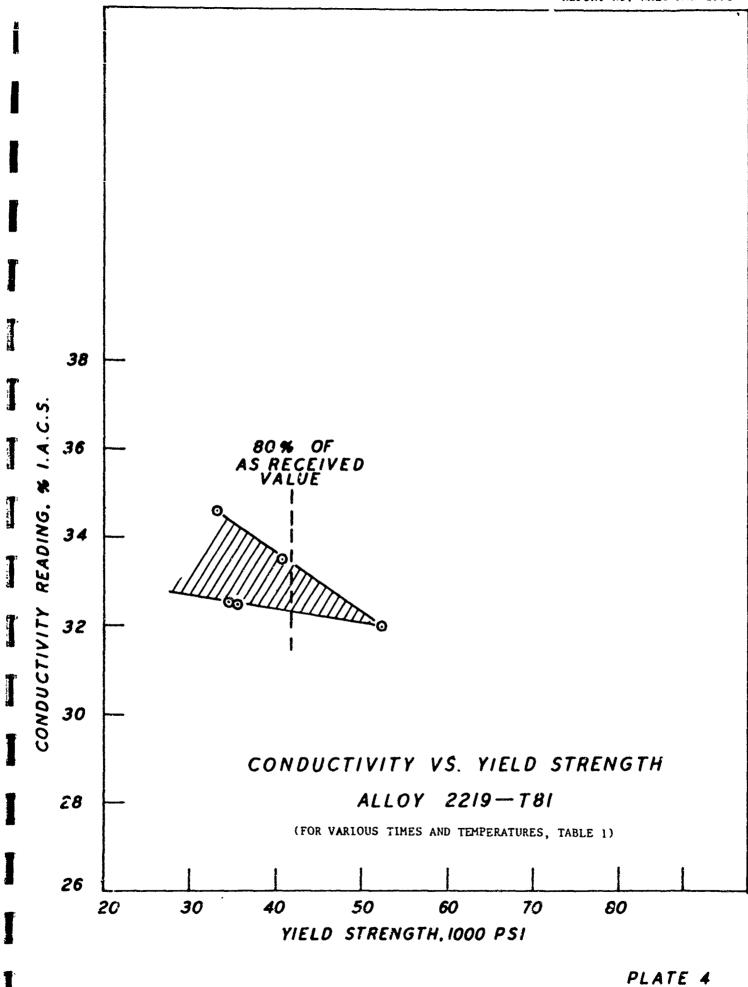
PLATE 3

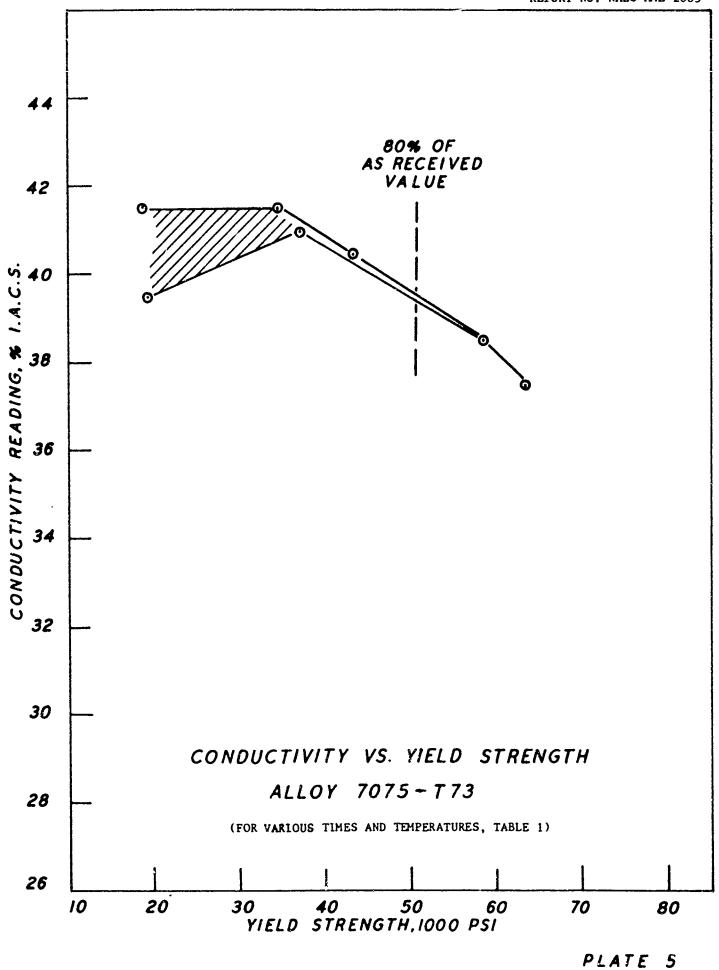
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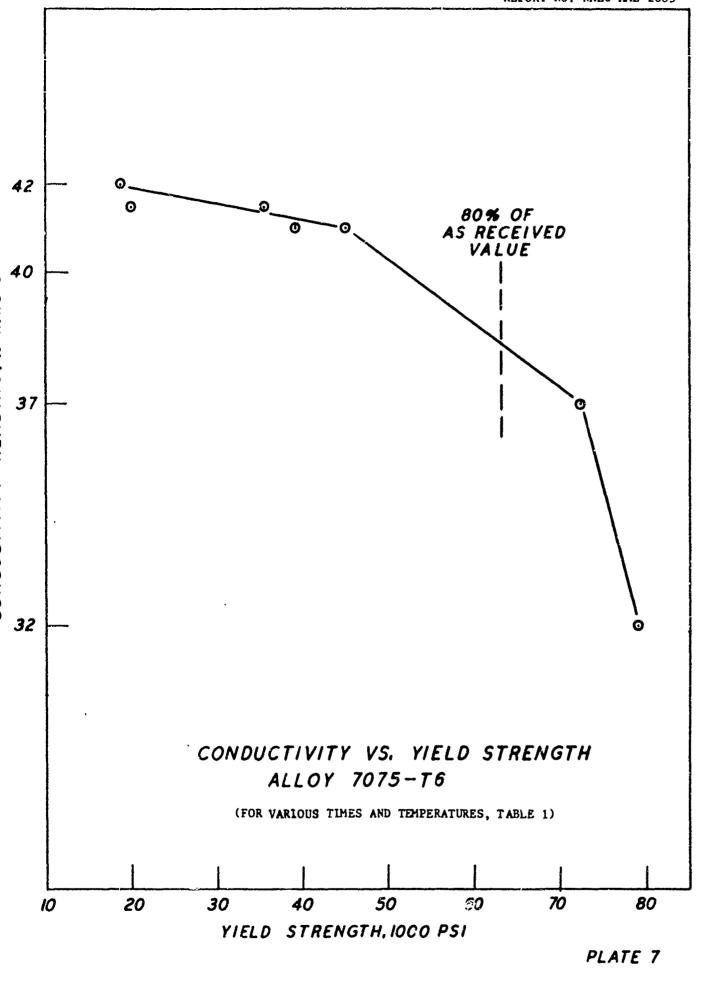
CONDUCTIVITY READING, % 1.A.C.S

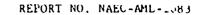
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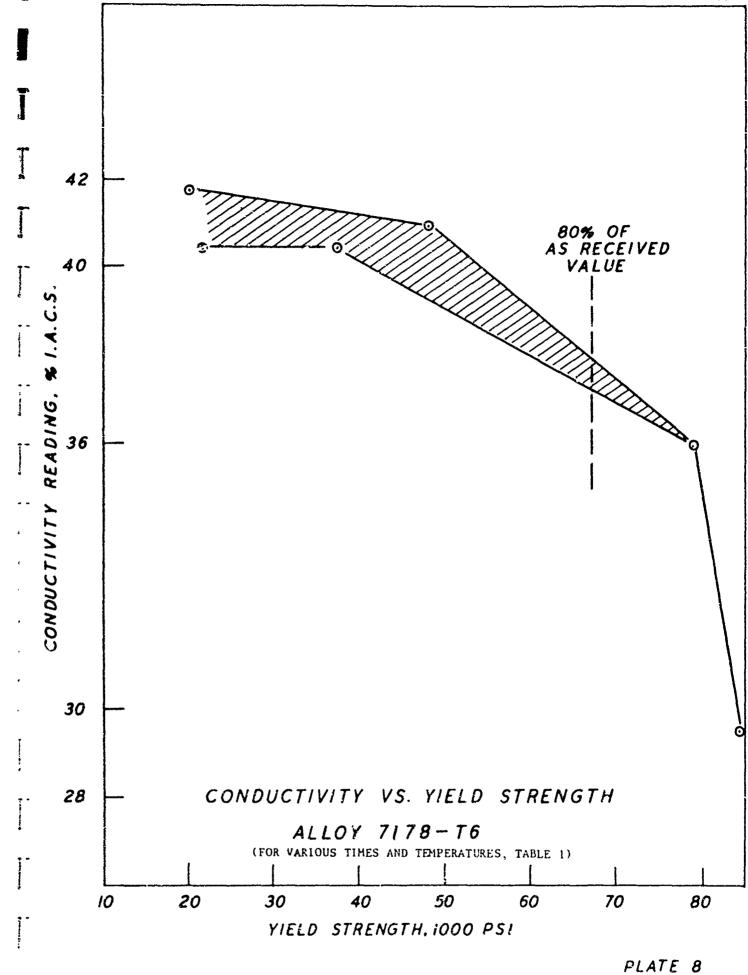
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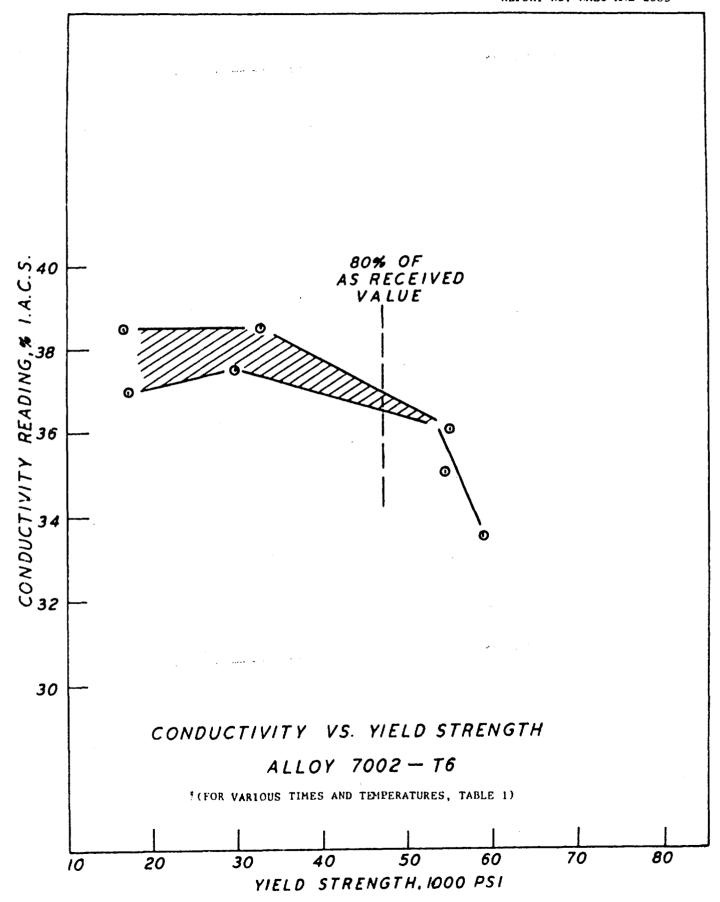


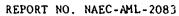


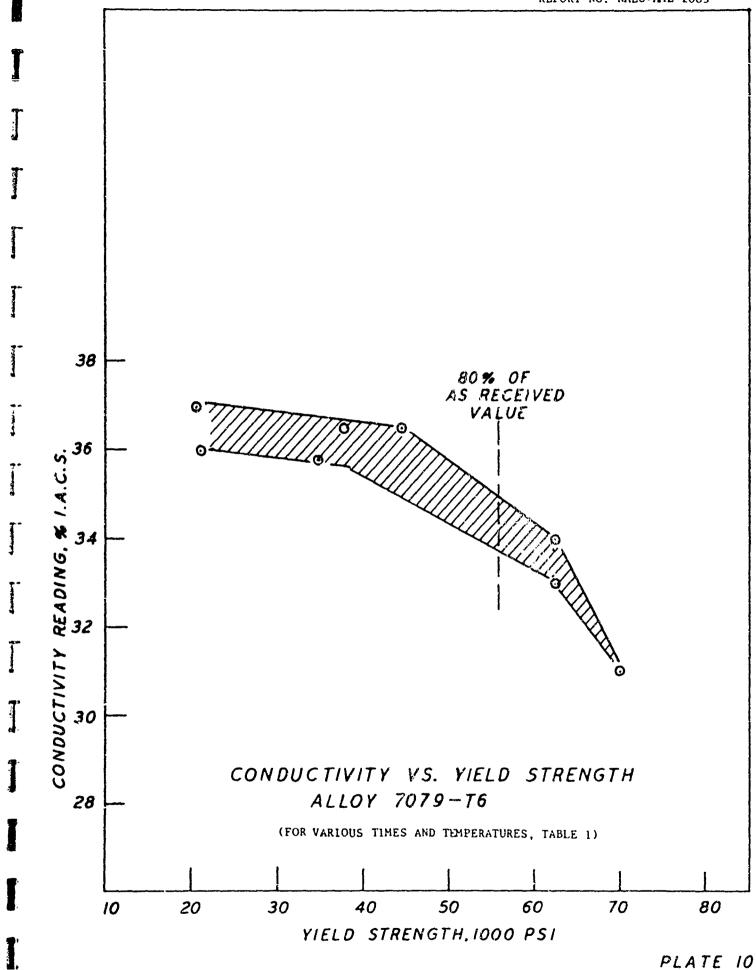












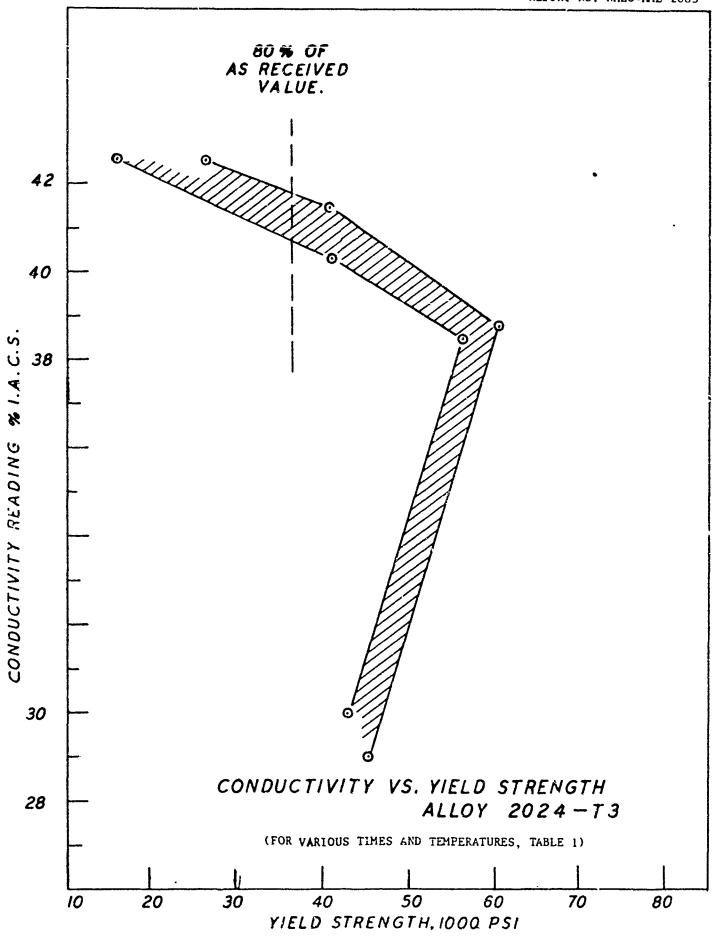


PLATE II

U. S. NAVAL AIR ENGINEERING CENTEP, PHILA, PA. 1. REPORT NO. NAEC-AML-2083 AERONAUTICAL MATERIALS LABORATORY	U. S. MAVAL ATR ENGINEERING CENTER, FHILA, PA. 1. REPORT NO. MAEG-AML-2083 AEROMAUTICAL MATERIALS LASONATORY (2. PAN 10-40
investigation into the Electrical Conductivity and Mechanical Properties of Aluminum Alloys Subjected to Elevated Temperature Exposure; W. Allan/R. Mahorter, December 1964,	Investigation into the Electrical Conductivity and Mechanical Properties of Aluminum Alloys Subjected to Elevated Temperature Exposure; W. Allen/R. Mahorter, December 1964,
The relationship between the electrical conductivity (as measured by Magnatester Conductivity Meter, F100 Series) and strength properties of bare aluminum alloys 7075-16, 7075-173, 6061-16, 7178-16, 7002-16, 2024-181, 2024-183, 7079-16, 2020-16, and 2219-181 was investigated in an attempt to correlate conductivity with heat damage to aircraft structural alloy.	The relationship between the electrical conductivity (as measured by Magnatester Conductivity Meter, F100 Series) and strength properties of bare alum:num alloys 7075-T6, 7075-T73, 6061-T6, 7178-T6, 7002-T6, 2024-T81, 2024-T83, 7079-T6, 2020-T6, and 2219-T81 was investigated in an attempt to correlate conductivity with heat damage to aircraft structural alloys.
U. S. MAVAL AIR ENGINEERING CENTER, PHILA, PA. 1. REPORT NO. MAEG-AML-2083 Aeromautical materials laboratory	U. S. NAVAL AIR ENGINEERING CENTER, PHILA., PA. 11. REPORT NO. NAEG-AML-2083 AERONAUTICAL MATERIALS LABORATORY [2, PAN 10-40]
Investigation into the Electrical Conductivity and Mechanical Properties of Aluminum Alloys Subjected to Elevated Temperature Exposure; W. Allen/R. Mahorter, December 1964, 2 Tables, 11 Piates	Investigation into the Electrical Conductivity and Mechanical Properties of Aluminum Alloys Subjected to Elevated Temperature Exposure; W. Allen/R. Maherter, December 1964,
The relationship between the electrical conductivity (as measured by Magnatester Gonductivity Meter, F100 Series) and strength properties of bare aluminum alloys 7075-76, 7075-173, 6061-76, 7178-76, 7002-16, 2024-781, 2024-783, 7079-76, 2020-76, and 2219-781 was investigated in an attempt to correlate conductivity with heat damage to aircraft structural alloys.	The relationship between the electrical conductivity (as pressured by Magnatester Conductivity Meter, F100 Series) and strength properties of bare aluminum alloys 7075-16, 7075-173, 6061-16, 7178-16, 7002-16, 2024-181, 2024-183, 7079-16, 2020-16, and 2210-181 was investigated in an attempt to correlate conductivity with heat demage to aircraft structural alloys.